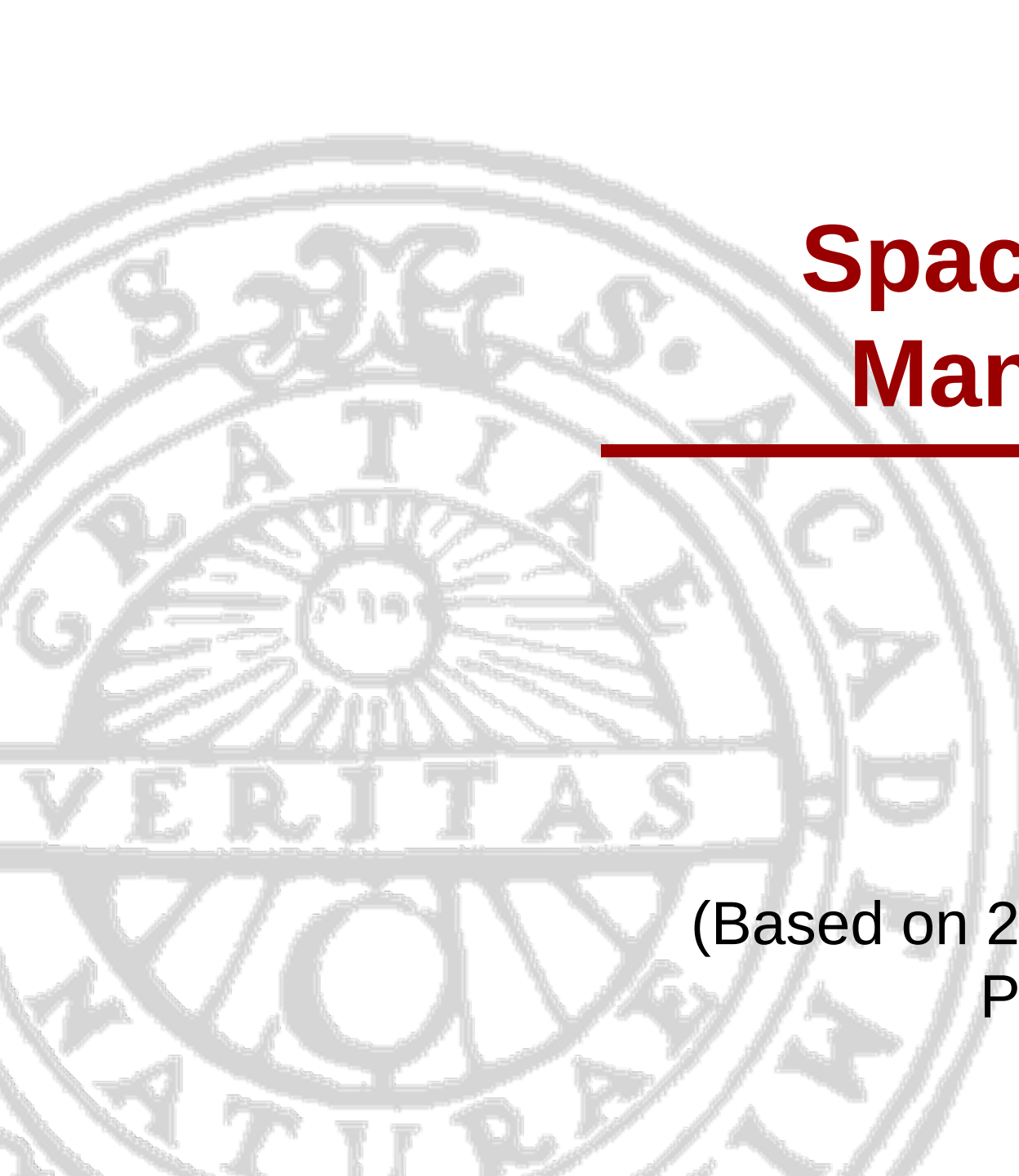




Space Project Management

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8 April 2008

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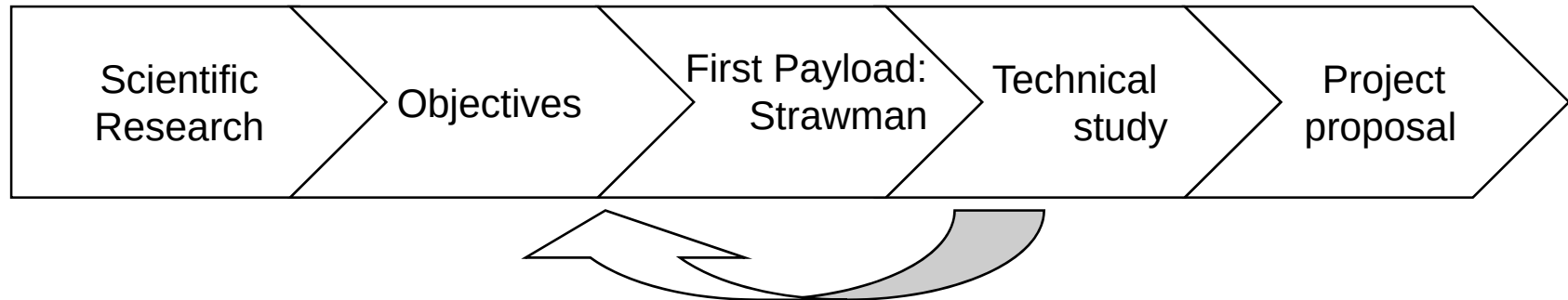
(Based on 2007 lectures by
Petrus Hyvönen)

Outline

- Objectives and requirements
- Project management tools
- ESA project phases
- General comments

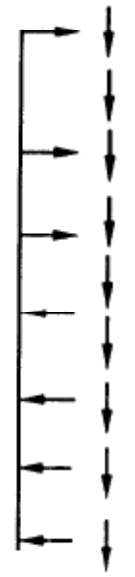
Mission Objectives

- The mission objectives define the goals and requirements of the mission
 - ✱ *To investigate Venus atmosphere and its coupling to the solar wind to understand the development of Venus atmosphere and hydrosphere*



(Note feedback all the way to the objectives)

Space Mission Analysis and Design Process

Typical Flow	Step	Section
	Define Objectives { A. Define broad objectives and constraints B. Estimate quantitative mission needs and requirements	1.3 1.4
	Characterize the Mission { C. Define alternative mission concepts D. Define alternative mission architectures E. Identify system drivers for each F. Characterize mission concepts and architectures	2.1 2.2 2.3 2.4
	Evaluate the Mission { G. Identify driving requirements H. Evaluate mission utility I. Define mission concept (baseline)	3.1 3.2 3.3
	Define Requirements { J. Define system requirements K. Allocate requirements to system elements	4.1 4.2-4.4

- Note that requirement is defining what is to be done, and not how
- Requirements are quantitative measurements of how well objectives are met

Top level mission requirements

Top Level Mission Requirements		
<u>Functional Requirements</u>	<u>Operational Requirements</u>	<u>Constraints</u>
Performance Coverage Responsiveness Secondary missions	Mission Duration Availability Timeliness Survivability Data Distribution Data Content Data Form Data Format	Cost Schedule Regulations Political limits Environment External Interfaces Development Constraints Spacecraft Disposal

Science objective vs Payload instrument matrix

Scientific Objective	Payload Instrument										
	Ion mass spectrometer	Ion and electron spectrometer	Plasma wave/lightning detector	Langmuir probe	Magnetometer	ENA spectrometer	Mass-resolving driftmeter	Near IR imaging spectrometer	UV/visible photometer	Sub-satellites	Plasma sounder
Venus atmosphere – solar wind interaction	X	X	X	X	X	X	X			X	
Direct measurement of escaping ions	X	X		X		X	X			X	
Water escape in the past	X	X									
Aurora		X	X	X					X		X
Acceleration processes	X	X	X	X	X	X	X			X	
Lightning			X						X		
Super-rotation	X	X			X		X	X			X
Greenhouse effect								X			X
Dawn-dusk asymmetry	X	X		X		X	X			X	
Meso-scale properties										X	
Magnetic field					X						
Volcanic activity	X	X		X			X	X	X		
Kinetic effects of the solar radiation	X	X					X				

Strawman Payload

- First payload that is used as a baseline in the design of the spacecraft
- Will be changed during a formal call
- Important to early get a design of the spacecraft and understand the requirements

<i>Instruments</i>	kg
Mass resolving ion spectrometer	5
Electron & ion spectrometer	2
Wave/lightning detector	2
Langmuir probe(s)	1
Magnetometer	0,5
ENA spectrometer	2,5
Driftmeter with mass resolution	3
Subsatellites	16
Subsatellite release/support system	3
<i>Atmospheric instruments</i>	
Near IR imaging spectrometer	10
UV/visible photometer	3
Radio science experiment	0
Total Mass:	48 kg

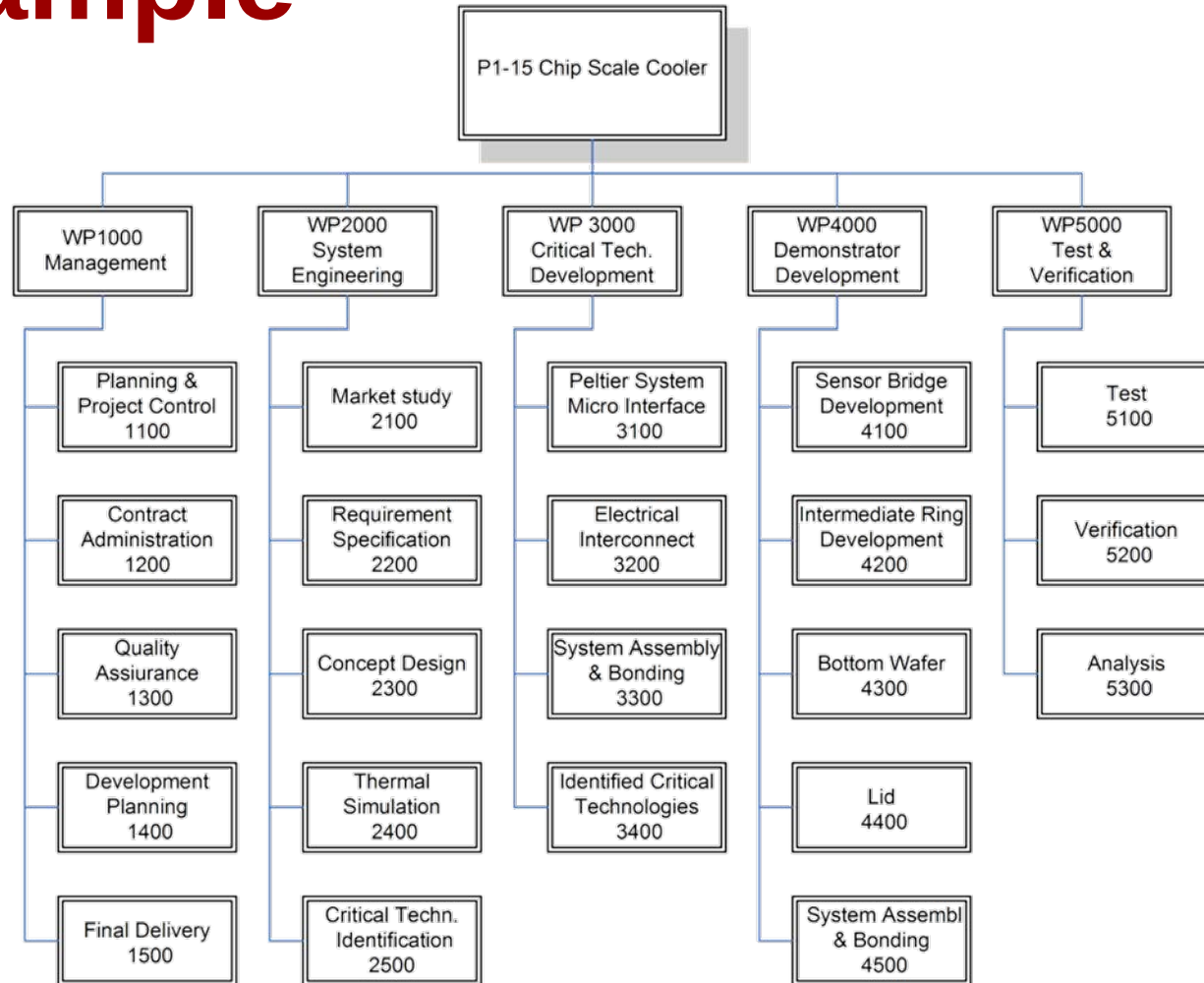
Project management tools

- Work breakdown structure
 - ✱ Total project work structured
- Gantt Chart
 - ✱ Work as a function of time with critical links
- Organizational chart
- Integrated models

Work package breakdown & Work package description

- On a large project, the work should be broken down into manageable subunits (work packages).
 - ✱ Tree structure ensures all work packages sum to 100% of task.
- A work package description:
 - ✱ is measurable and manageable in its scope
 - ✱ is allocated to only one manager
 - ✱ results in supply of products or documents
 - ✱ identifies all inputs and outputs, including interfaces with other tasks or WPs
 - ✱ clearly identifies planning constraints

Work package breakdown – example



Work package description – example

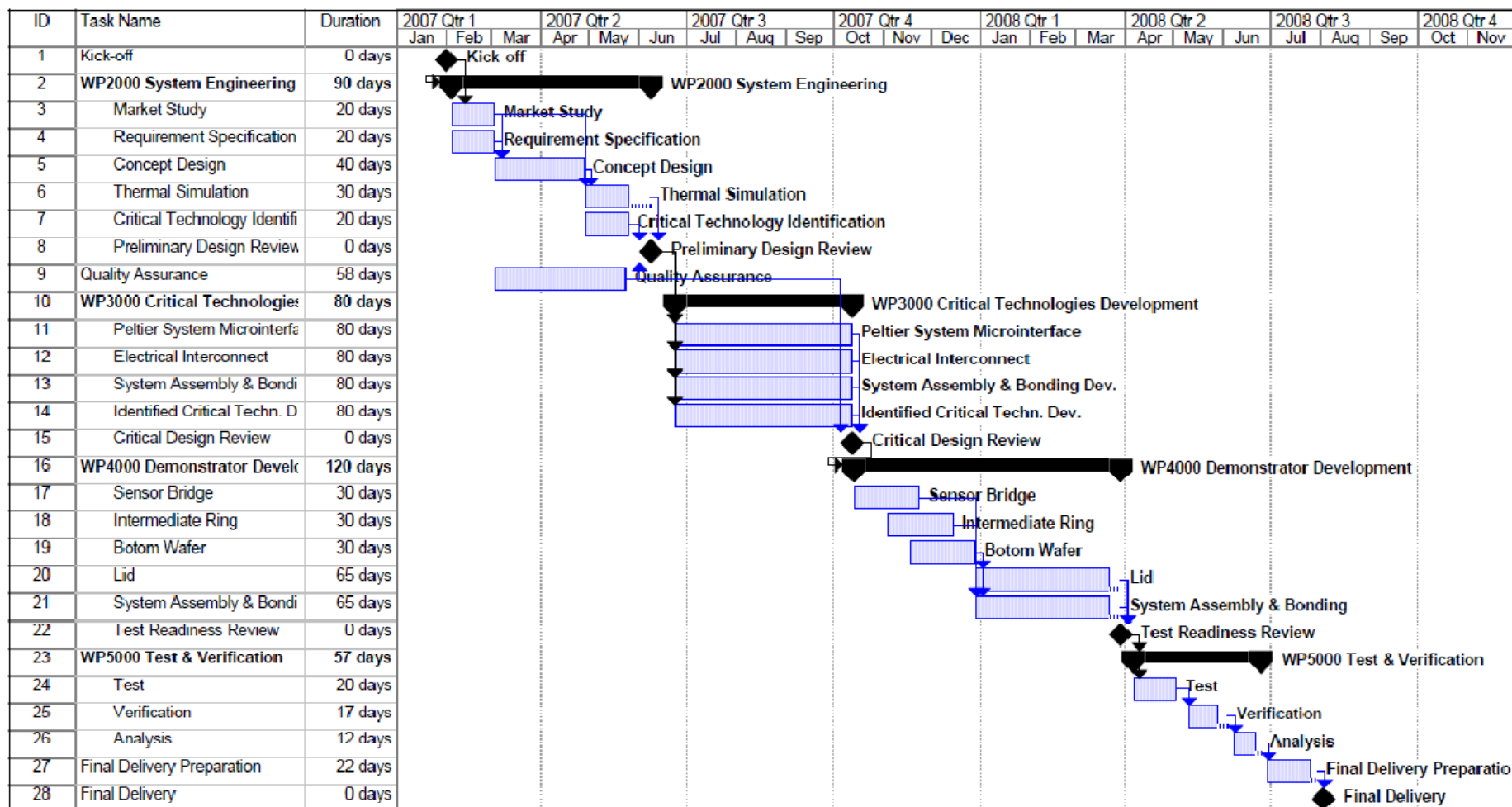
PROJECT: CHIP SCALE COOLERS		W.P. REF: 1400
W.P. TITLE: Development Planning		SHEET 1 OF 1
CONTRACTOR:		ISSUE NO: 1
MAJOR CONTST: System Engineering		ISSUE DATE: 2006-09-21
START EVENT: Kick-off	PLANNED DATE:	
END EVENT: Final delivery	PLANNED DATE:	
W.P. MANAGER: Petrus Hyvönen		
OBJECTIVES: To establish and maintain development plans for the chip-scale cooler demonstrator		
INPUT REQUIRED TO START: – Kick-off or authorization to start – Authorization to proceed with proposed objectives		
TASKS INCLUDED: – Establish the development plan for chip-scale coolers – Maintain and update the development in a controlled way through the development process – Compilation of test and verification plan		
OUTPUT: – Detailed development plan for chip-scale coolers – Contribution to design reviews and final report		

Schedule

As a minimum, the baseline schedule contains:

- key milestones
- descriptions of the activities
- start and finish dates of activities
- duration of activities
- identification of the critical path activities

Gantt Chart



Project: P1-15 Gantt.mpp
Date: 2006-09-07

Task



Milestone



External Tasks



Split



Summary



External Milestone



Progress



Project Summary

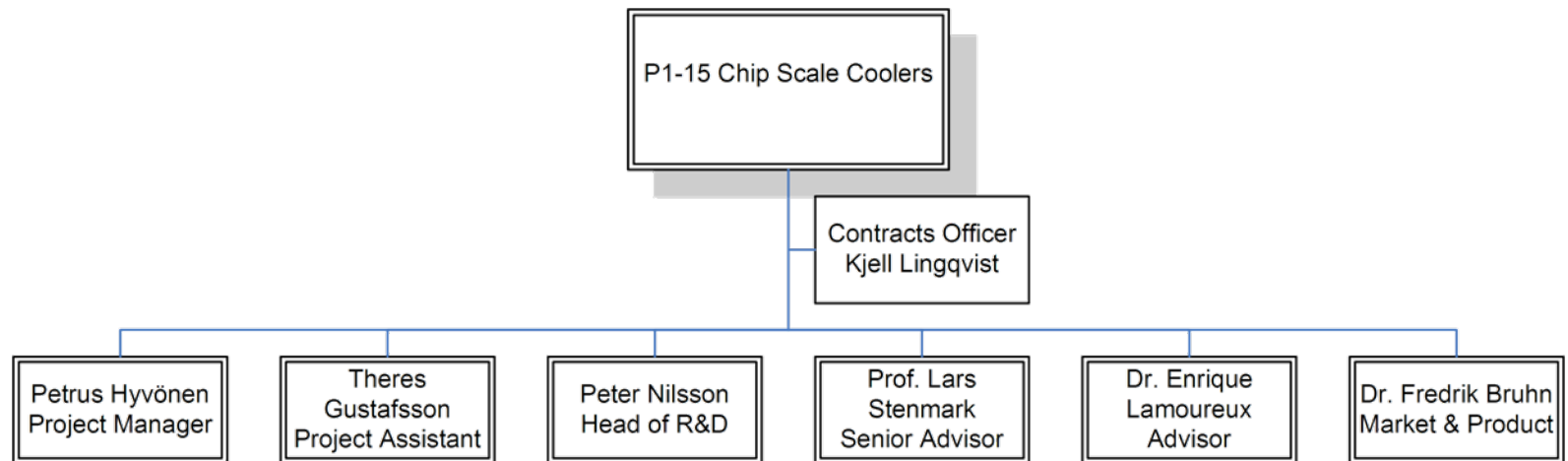


Deadline



Organizational Chart

- Clear and unambiguous definition of individual roles, responsibilities and authority
- Both internal and external to the organization



Integrated Models & Concurrent engineering

- Concurrent engineering is a method to make early concept designs
 - ✱ "Gather all in same room, have a boss, and make sure it all agrees in a semi-formal way"
- Tested by ESA, NASA and many universities with successful results at early project stages
- Integrated models are a central tool in this method
 - ✱ Collect and link calculations, values and other information
 - ✱ Enable all to access, modify and inspect
 - ✱ Uses COTS software – Excel
 - Commercial Off The Shelf

Integrated Model Example

Microsoft Excel - Venus Express IM.xls

File Edit View Insert Format Tools Data Window Help

Geneva 10 B I U

G22

	A	B	C	D	E	F	G	H	I	J	K	L	M
1	Venus Express Power												
2		Page version	0	Last modified	6/19/2005								
3													
4	Page and Author Info												
11		<i>Press Gadget to Expand/Hide info</i>											
12													
13	Input Parameters					Output Parameters			Colour encoding:				
14		Distance From Sun	0.7233 AU			Available Power	401 W				Green	Inmovable constant	
15		Solar Intensity at 1 AU	1358 W/m^2								Blue	Input value	
16											Red	Warnings and Loop closures	
17													
18													
19		<i>No crosspage links may be made under this line!</i>											
20													
21													
22	Solar Intensity												
23		Distance from sun	0.7233 AU										
24		Intensity at Earth	1358 W/m^2										
25		Intensity at Distance	2596 W/m^2										
26													
27	Solar Panels												
28		Spacecraft diameter	2.3 m										
29		Spacecraft Height	0.8 m										
30		Panel Cross Section	1.84 m^2										
31													
32	Power Generation												
33		Total Panel Area	1.84 m^2										
34		SolarConstant at Earth distance	2596 W/m^2										
35		Coverage ratio	0.6										
36		Cell Efficiency	14%										
37		Silicon	14%	Silicon		From SMAD							
38		GaAs	18%	GaAs		From SMAD							
39		Multilayer	26%			Techstar							
40		Total Panel Efficiency	8%										
41		Total Solar Power	401 W										
42													
43													
44													
45													
46													

Sheet1 Mars Properties Orbital Motion Venus Properties Interpl. Com S-LGA TM S-HGA TM X-TM S-LGA TC S-HGA TC MD Instrumentation

Simple version handling

Structure

Possibility to define output / input variables

Colorcoding for defining value properties

References to external sources

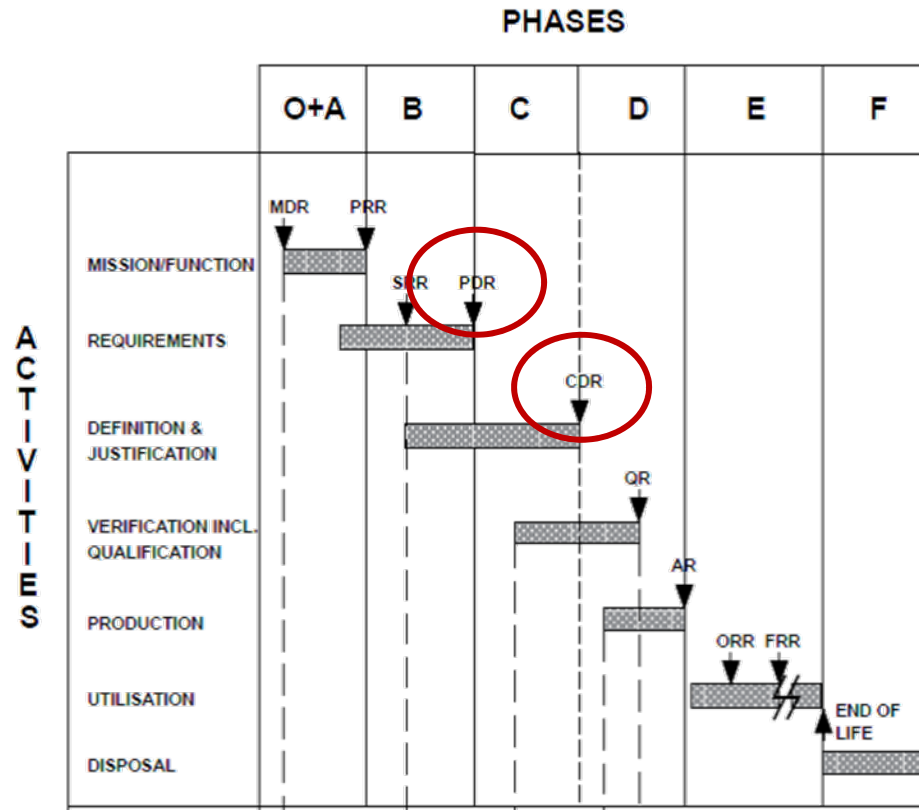
Reusability

Tabs for different subsystems

Cost estimation

- Cost estimation needs to rely on good knowledge or good assumptions
 - ✱ Use previous missions and make a range of what seems reasonable
- Parametric model exists for common types of spacecraft
 - ✱ The applicability of these can be discussed
 - ✱ SMAD has a simple model for small satellite cost estimation

Project phases – ESA project



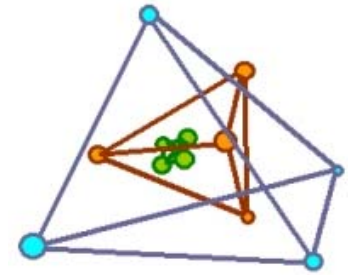
NOTE

- AR = Acceptance Review
- CDR = Critical Design Review
- FRR = Flight Readiness Review
- MDR = Mission Definition Review
- ORR = Operational Readiness Review
- PDR = Preliminary Design Review
- PRR = Preliminary Requirements Review
- QR = Qualification Review
- SRR = System Requirements Review
- WBS = Work Breakdown Structure

- A phase is a group of activities
 - ✱ advances a project from one milestone to another
 - ✱ often ended by a formal review that confirms if the work has been done follows the requirements

Phase 0: Mission Analysis and Needs Identification

- (NASA terminology: pre-phase A)
 - Identification and characterization
 - Expected performance
 - Assessment of operating constraints
 - ✱ in particular: physical and operational environment,
 - Identification of possible system concepts
 - preliminary assessment of project management data (organization, costs, schedules)
 - End result: specification of the mission concept
-
- At the end of phase 0, a Mission Definition Review (MDR) can take place
 - ✱ (NASA terminology: Mission Concept Review, MCR)

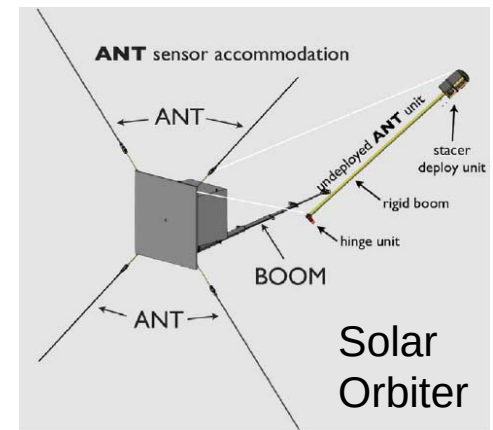


Cross-scale

Phase A: Feasibility

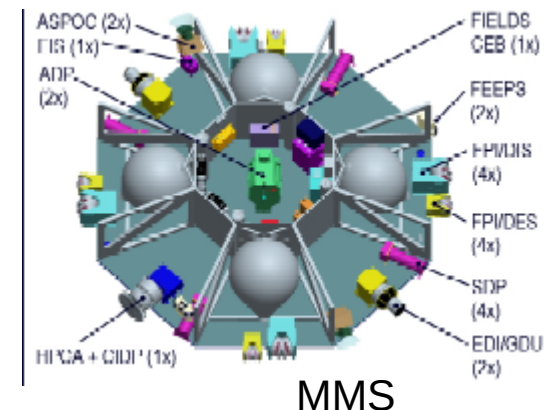
Finalize needs expressed in phase 0 and propose solutions

- Quantify and characterise critical elements for technical and economic suitability
 - ✱ explore various possible system concepts
 - ✱ compare these concepts against the needs
 - determine levels of uncertainty and risks
 - ✱ estimate the technical and industrial feasibility,
- Identify constraints:
 - ✱ costs, schedules, organisation, utilisation (operations, implementation, maintenance),
- At the end of the phase A, the Preliminary Requirements Review (PRR) is conducted
 - ✱ (NASA terminology: Mission Definition Review, MDR)



Phase B: Preliminary Definition (Project and Product)

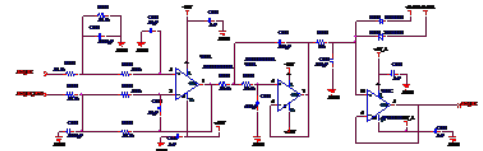
- Select technical solutions for system concept selected in phase A
 - ✱ acquire a precise and coherent definition (performance levels, costs, schedules) at every level
- System Requirements Review (SRR)
 - ✱ Initial identification of 'Make or buy' alternatives
- Confirm feasibility of the recommended solution
- At the end of phase B:
 - ✱ Preliminary Design Review



Phase C: Detailed Definition

- Detailed study of phase B solution
 - ✱ production of representative elements
 - ✱ detailed definition of system and components
- Allows a definitive 'make or buy' decision
- Allows confirmation of the test and qualification setup
- Allows preparation of phase E activities.

At the end of this phase, the Critical Design Review (CDR) is conducted.



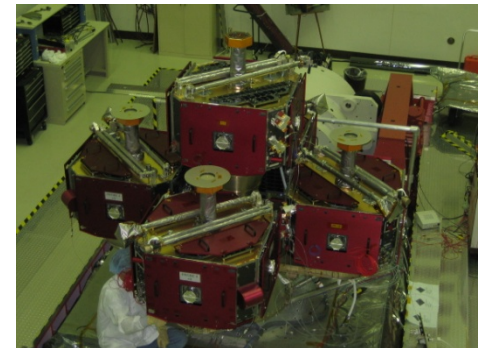
Phase D: Production/Ground Qualification Testing

Phase D is the end of the system development.

- Ground qualification testing verifies:
 - ✱ technical conformity of the components against the requirements (**design qualification**)
 - ✱ aptitude to be used operationally (**operational qualification**)
 - ✱ identification of the functional and operational margins.

Phase D ends with the Acceptance Review (AR)

Phases C and D are usually inseparable



Phase E: Utilisation

- Launch campaign, launch and in-flight acceptance of space elements
- Operation and maintenance of the system, feedback.
- Can be divided into two sub-phases:
 - ✱ E1: test and commissioning
 - ✱ E2: utilisation



This course

- Designing to a pre-phase 0 level
 - ✱ Don't need, e.g., detailed circuit diagram of electronics
 - ✱ Do need to demonstrate that the system could be built, and what it would be able to do if it were built
- Get organized!
 - ✱ Define roles and responsibilities
 - ✱ Have a plan for how to meet deadlines
 - ✱ Don't get too far ahead of yourselves

This course

■ "Phases"

✱ Gathering questions

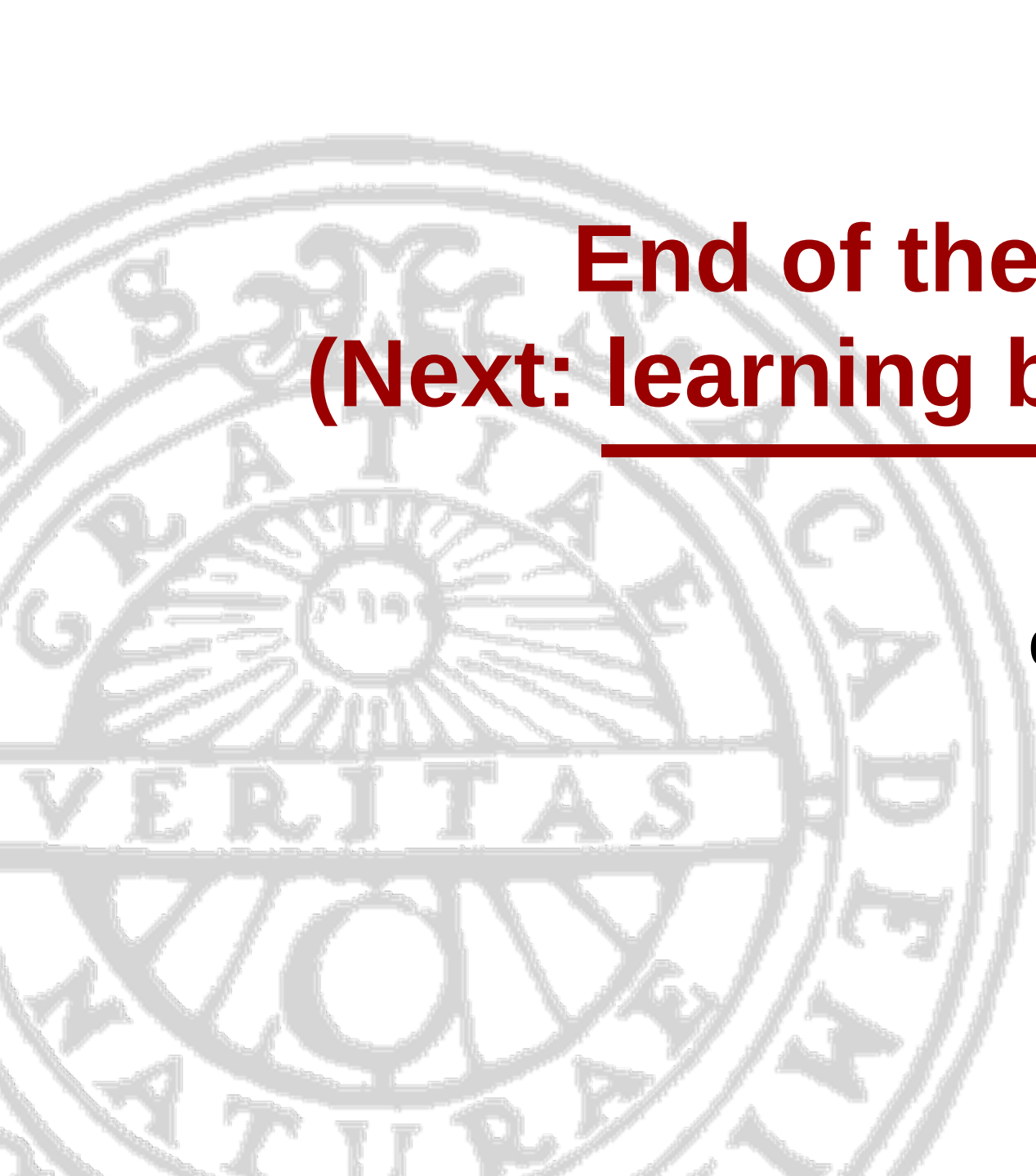
- Meeting with experts: week of April 21

✱ Preliminary strawman payload

- Draft chapters May 16

✱ Iterating

- making into a consistent story
- Making "sellable"



End of the lectures (Next: learning by doing)

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